#### Case C3.6: Flow over a 2D periodic hill DNS results at Re = 2,800

D. Moxey, J. Peiro, S. Sherwin

Department of Aeronautics, Imperial College London

2nd International Workshop on High-Order CFD Methods Cologne, 28th May 2013

## Overview

- Outline of numerical discretisation
- Representative results
- Summary

- Joint continuous discretisation: 2D nodal spectral element (quads only), 1D Fourier pseudospectral.
- ► High-order velocity correction discretisation of incompressible Navier-Stokes equations.
- Three-step splitting scheme: explicit advection, implicit pressure/velocity solves.
- Direct static-condensation on each 2D plane.
- ► Volumetric flowrate imposed through Stokes correction function.
- ► Parallelisation in Fourier direction using MPI (expensive!)

- Joint continuous discretisation: 2D nodal spectral element (quads only), 1D Fourier pseudospectral.
- High-order velocity correction discretisation of incompressible Navier-Stokes equations.
- Three-step splitting scheme: explicit advection, implicit pressure/velocity solves.
- Direct static-condensation on each 2D plane.
- ► Volumetric flowrate imposed through Stokes correction function.
- ► Parallelisation in Fourier direction using MPI (expensive!)

- Joint continuous discretisation: 2D nodal spectral element (quads only), 1D Fourier pseudospectral.
- ► High-order velocity correction discretisation of incompressible Navier-Stokes equations.
- Three-step splitting scheme: explicit advection, implicit pressure/velocity solves.
- Direct static-condensation on each 2D plane.
- ► Volumetric flowrate imposed through Stokes correction function.
- ► Parallelisation in Fourier direction using MPI (expensive!)

- Joint continuous discretisation: 2D nodal spectral element (quads only), 1D Fourier pseudospectral.
- ► High-order velocity correction discretisation of incompressible Navier-Stokes equations.
- Three-step splitting scheme: explicit advection, implicit pressure/velocity solves.
- Direct static-condensation on each 2D plane.
- ► Volumetric flowrate imposed through Stokes correction function.
- ► Parallelisation in Fourier direction using MPI (expensive!)

- Joint continuous discretisation: 2D nodal spectral element (quads only), 1D Fourier pseudospectral.
- ► High-order velocity correction discretisation of incompressible Navier-Stokes equations.
- Three-step splitting scheme: explicit advection, implicit pressure/velocity solves.
- Direct static-condensation on each 2D plane.
- Volumetric flowrate imposed through Stokes correction function.
- ► Parallelisation in Fourier direction using MPI (expensive!)

- Joint continuous discretisation: 2D nodal spectral element (quads only), 1D Fourier pseudospectral.
- ► High-order velocity correction discretisation of incompressible Navier-Stokes equations.
- Three-step splitting scheme: explicit advection, implicit pressure/velocity solves.
- Direct static-condensation on each 2D plane.
- ► Volumetric flowrate imposed through Stokes correction function.
- Parallelisation in Fourier direction using MPI (expensive!)

### Simulation resolution



- Very sensitive to choice of resolution
- > 2D mesh: 3,626 elements, polynomial order P = 6.
- 160 Fourier planes
- $\blacktriangleright$  ~ 28 million degrees of freedom
- Custom mesh generation

## Simulation resolution



- Very sensitive to choice of resolution
- > 2D mesh: 3,626 elements, polynomial order P = 6.
- 160 Fourier planes
- $\blacktriangleright$  ~ 28 million degrees of freedom
- Custom mesh generation

#### Simulation parameters

Parameter	Value
$\Delta t$ $T_{avg}$ $Re_b$ $N_{proc}$ $T_{Table}$	0.001 <i>t<sub>c</sub></i> 275 <i>t<sub>c</sub></i> 2,800 80 9.281s

- Simulation uses around 32k work units per t<sub>c</sub> (~ 12 days for T<sub>avg</sub>).
- ▶ Normalisation by bump height *h* and bulk velocity *u<sub>b</sub>*.
- Convective time  $t_c := tu_b/h$

#### Flow features



# Velocity profiles



# Velocity profiles: comparison

Comparison of simulation vs. reference solutions (LESOCC), available on QNET CFD wiki. At x/h = 0.05:



Good convergence at inlet

# Velocity profiles: comparison

Comparison of simulation vs. reference solutions (LESOCC), available on QNET CFD wiki. At x/h = 2:



Not so great in recirculation region (time-averaging/resolution)

# Velocity profiles: comparison

Comparison of simulation vs. reference solutions (LESOCC), available on QNET CFD wiki. At x/h = 8:



Improves as we continue downstream

#### Reynolds stresses



#### Reynolds stresses: comparison

Comparison of simulation vs. reference for Reynolds stresses show a similar picture. At x/h = 0.05:



#### Reynolds stresses: comparison

Comparison of simulation vs. reference for Reynolds stresses show a similar picture. At x/h = 2:



#### Reynolds stresses: comparison

Comparison of simulation vs. reference for Reynolds stresses show a similar picture. At x/h = 8:



## Summary

- Results show fairly good agreement with existing linear profiles...
- ... however still probably requires further temporal averaging for statistical convergence.

Future work:

- Investigate meshing options to reduce number of degrees of freedom required.
- Comparison against experimental data at Re = 5,600.
- Implement parallelism in spectral element planes for more efficient communication pattern (Nektar++).